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Aerodynamics (2) Stability and Control (1)

IN INIA6

Airplanes, - Control characteristics (06395); Airplanes - Stability (06167); Airplanes, Bomber - Performance charts (06517)

Charts showing stability and control characteristics of sirplanes in flight

National Advisory Committee for Aeronautice, Washington, D. C.

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NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

WARTIME REPORT

ORIGINALLY ISSUED
December 1944 as
Memorandum Report 14126

CHARTS SHOWING STABILITY AND CONTROL CHARACTERISTICS

OF ATRPLANES IN FLICRE

By Stability and Control Section of Flight Research Division

Langley Memorial Aeronautical Laboratory Langley Field, Va.

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NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

MEMORANDUM REPORT

for the

Army Air Forces, Air Technical Service Command

and the

Bureau of Aeronautics, Navy Department

CHARTS SHOWING STABILITY AND CONTROL CHARACTERISTICS

OF AIRPLANES IN FLIGHT

By Stability and Control Section of Flight Research Division

INTRODUCTION

During October 1944, the National Advisory Committee conducted a series of conferences with the Army, Navy, and representatives of the aircraft industry for the purpose of discussing the flight-test procedures used in measuring the stability and control characteristics of airclanes. The conferences were initiated by the Army Air Forces, Air Technical Service Command, to acquaint the flight organizations of the industry with the flight test methods employed by the NACA and to standardize the techniques insofar as possible as they are employed by the various manufacturers and agencies engaged in determining the flying qualities of airplanes.

To facilitate the discussion during the conferences a series of charts was presented which portrayed typical good and undesirable airpline characteristics as determined in flight. The discussion centered around the characteristics portrayed and their relation to the Army Air Forces specifications for the stability and control of airplanes (reference 1). In general the following points were covered in connection with each chart:

- (a) The purpose of the test
- (b) The flight technique used

- (c) Items recorded
- (d) Evaluation and interpretation of data obtained

There were many requests from the conferees for copies of the charts presented for their further study and for reference with their notes taken during the discussion. Accordingly, the charts have been reproduced and are presented herein.

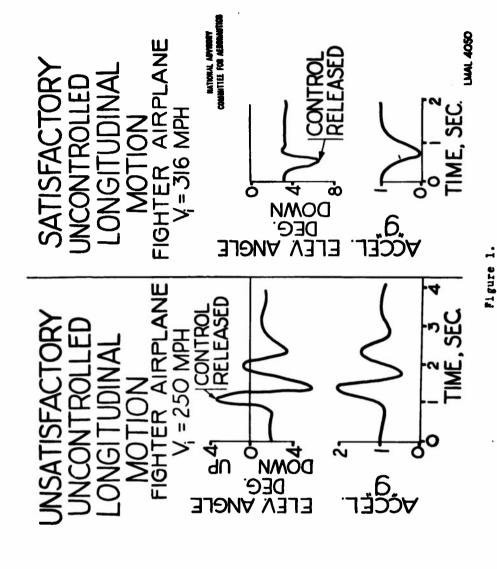
During the discussion there were many additional explanatory figures drawn on the blackboard so that the more formal charts do not give a complete picture of the material presented and discussed.

The conferences were held at both the Langley and Ames Laboratories of the NACA and a separate series of charts was presented by each group. Because of the similarity of the charts, however, only the charts presented at the Langley conference are given herein.

Langley Memorial Aeronautical Laboratory, National Advisory Committee for Aeronautics, Langley Field, Va., December 26, 1944.

REFERENCE

1. Anon.: Stability and Control Requirements for Airplanes. AAF Specification No. C-1815, Aug. 31, 1943.



SATISFACTORY STICK FIXED AND STICK FREE STATIC LONGITUDINAL STABILITY

FIGHTER AIRPLANE CLIMBING CONDITION

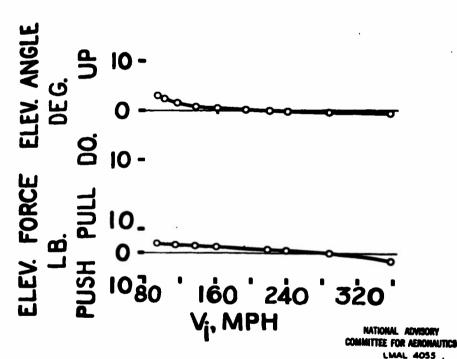
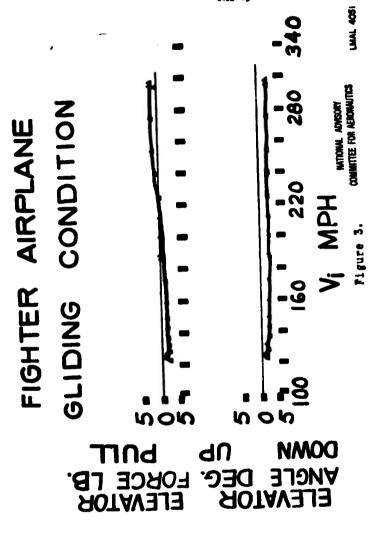


Figure 2.

UNSATISFACTORY STICK FREE LONGITUDINAL STABILITY



UNSATISFACTORY STICK FREE LONGITUDINAL STABILITY

FIGHTER AIRPLANE CLIMBING CONDITION RATED POWER, CLEAN

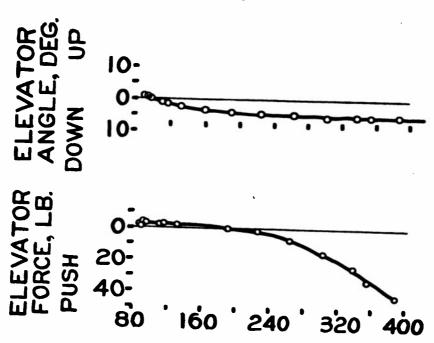


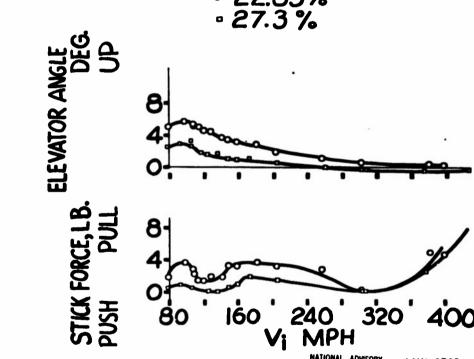
Figure 4. NATIONAL ADVISORY

UNSATISFACTORY STICK FREE LONGITUDINAL STABILITY

FIGHTER AIRPLANE
CLEAN CONDITION, POWER ON
C.G. POSITION IN % M.A.C.

• 22.85%

• 27.3%



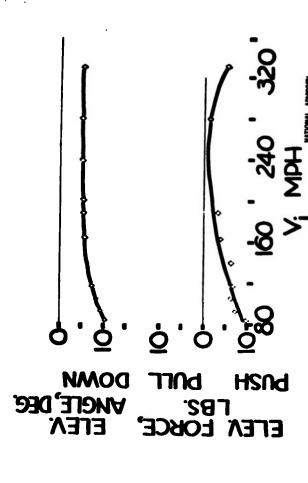
NATIONAL ADVISORY LIMAL 404
COMMITTEE FOR AERONAUTICS

Figure 5.

1-706

UNSATISFACTORY STICK-FIXED LONGITUDINAL STABILITY

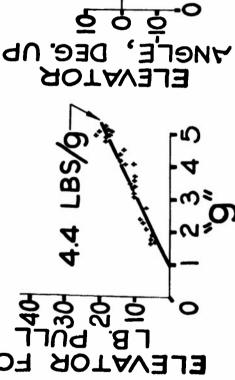
NAVY SCOUT BOMBER CLIMBING CONDITION



CTORY ELEVATOR CHARACTERISTICS ACCELERATED SATISFACTORY FORCE CHARAC Z

CONDITION FIGHTER AIRPLANE CRUISING

T LOBCE

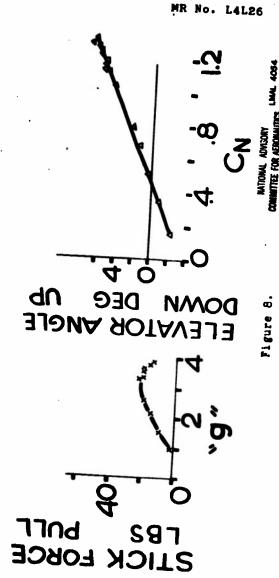


No. L4L26

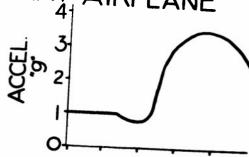
0.4 8 1.2 1.6 2.0

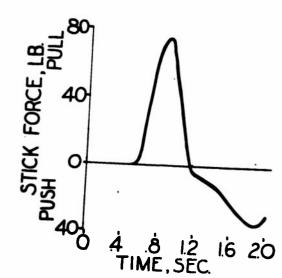
UNSATISFACTORY STICK FORCE CHARACTERISTICS IN ACCELERATED FLIGHT

NAVY SCOUT BOMBER WITH EXP BALANCE TAB V;=170 MPH





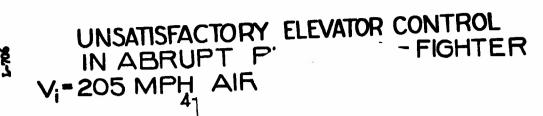




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Figure 9.



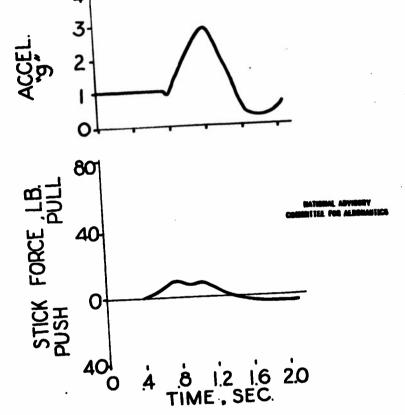


Figure 10.

LMAL 4045

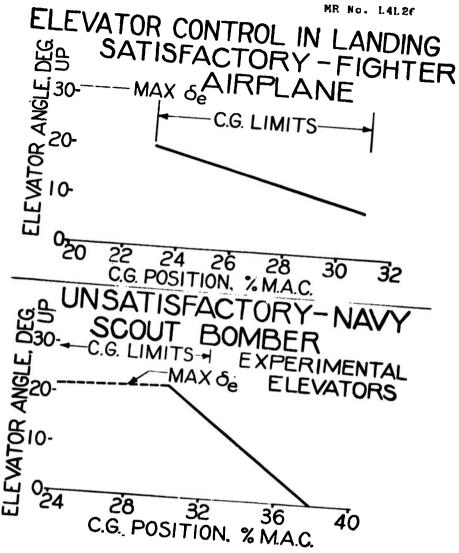


Figure 11.

LMAL 4046

CONDITION AT Vi=120 MPH

CHANGE IN ELEVATOR FORCE FROM CLIMBING

MR.No. L4L2

Figure 12.

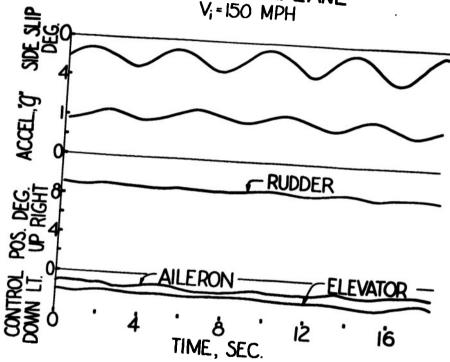
NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

MAI. 4041

MR No. L4L26

UNSATISFACTORY CONTROL FIXED OSCILLATION

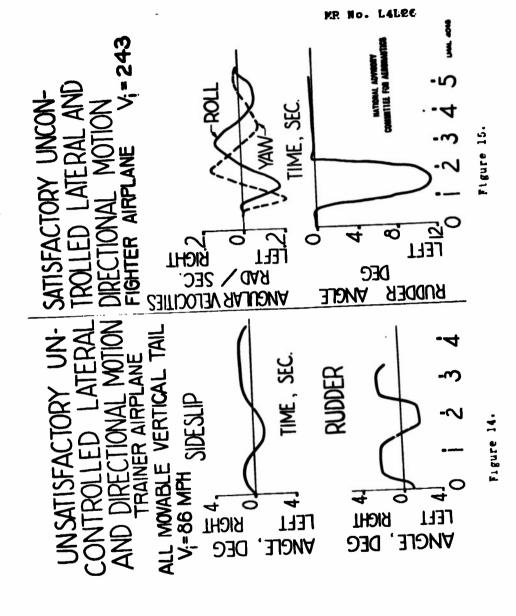
FIGHTER AIRPLANE V;=150 MPH

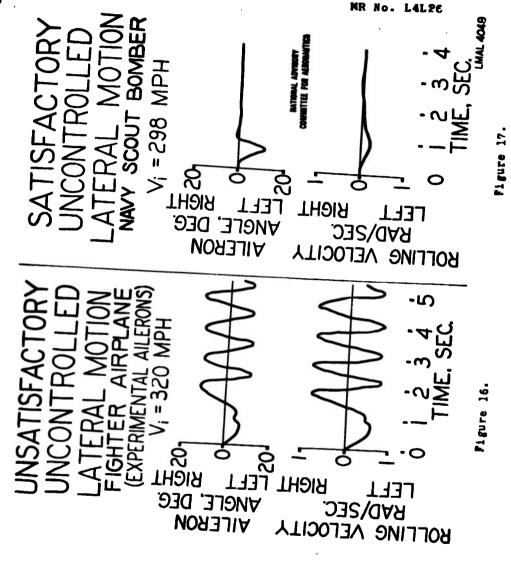


LMAL 4047

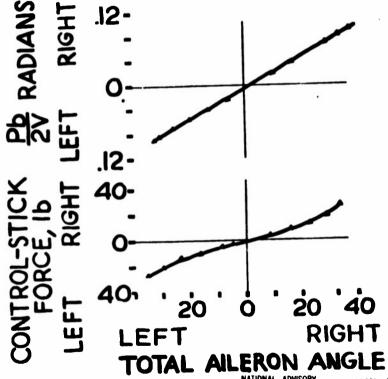
MATIONAL APPROPRY
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Figure 13





SATISFACTORY AILERON FORCE CHACTERISTICS FIGHTER AIRPLANE (EXPERIMENTAL AILERON) CLEAN CONDITION Vi = 201 MPH SNEW 11.12-



NATIONAL ADVISORY

LMAL 4040

Figure 18.

COMMITTEE FOR AERONAUTICS

UNSATISFACTORY AILERON FORCE CHARACTERISTICS FIGHTER AIRPLANE (CLEAN CONDITION) V; = 210 MPH

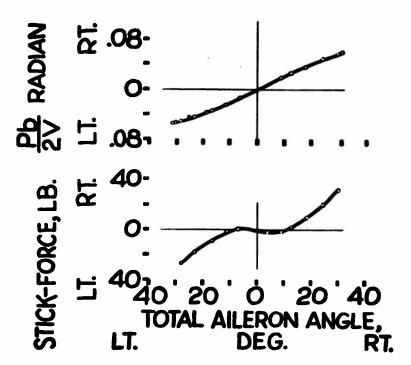
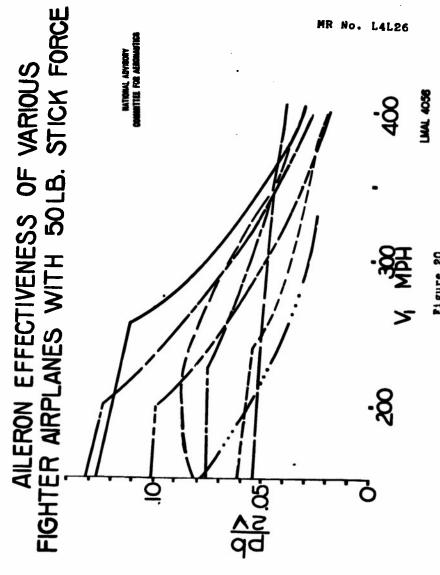


Figure 19. NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS LMAL 403:





UNSATISFACTORY DIRECTIONAL TRIM CHARACTERISTICS

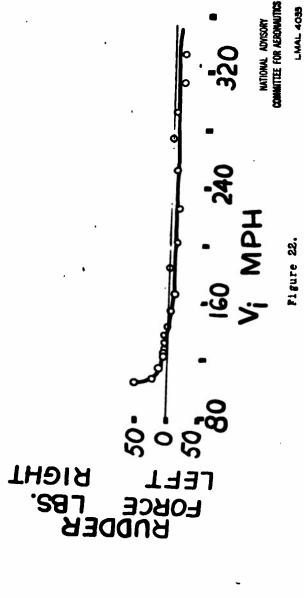
RATED POWER, CLEAN CONDITION FIGHTER AIRPLANE

COMMITTEE FOR AERONAUTIC 300 ECT 100-DECT 200- 100 140 180 220 260 STATE 200- 100 140 180 220 260 PEDAL RIGHT, RIGHT, 7 80 7

Figure 21.

SATISFACTORY DIRECTIONAL TRIM CHARACTERISTICS

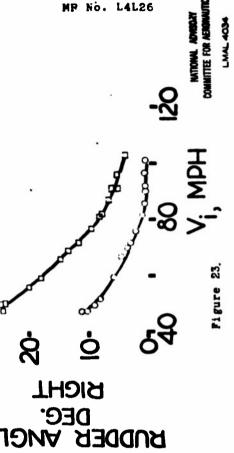
ATTACK BOMBER RATED POWER, CLEAN CONDITION



OF LATERAL C.G. RUDDER DEFLECTIC H N N IRED EFFECT HFT ON

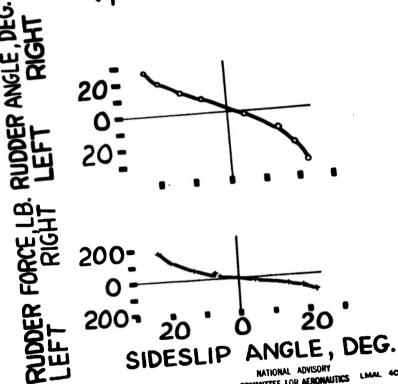


C.G. 4.16 IN. TO LEFT OF € C.G. 4.16 IN. TO RIGHT OF €



SATISFACTORY DIRECTIONAL STABILITY CHARACTERISTICS

FIGHTER AIRPLANE CLIMBING CONDITION Vi=90 MPH



UNSATISFACTORY RUDDER FREE STABILITY NAVY SCOUT BOMBER, CLIMBING CONDITION, V; = 120 MPH

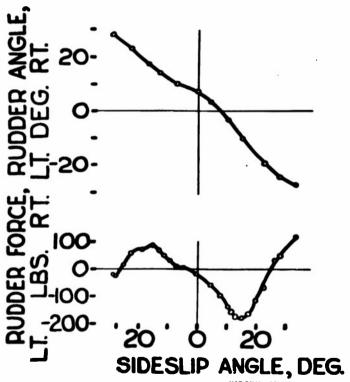


Figure 25. COMMITTEE FOR AERONAUTICS

JNSATISFACTORY RUDDER FIXED STABILITY

NAVY TORPEDO BOMBER, GLIDING CONDITION V; = 92 MPH

RUDDER ANGLE,
LT DEG. RT.
S, O, S

LT. ANGLE OF SIDESLIP, DEG. R

gure 26.

NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS LIMAL 4054

SATISFACTORY STICK FIXED AND STICK FREE DIHEDRAL EFFECT

FIGHTER AIRPLANE CLIMBING CONDITION, 252 MPH

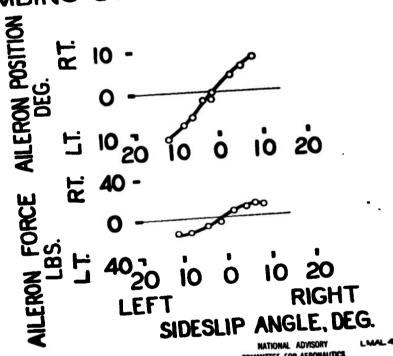


Figure 27.

COMMITTEE FOR AERONAUTICS

JNSATISFACTORY STICK FIXED DIHEDRAL EFFECT

FIGHTER AIRPLANE

LANDING CONDITION

V; = 120 MPH

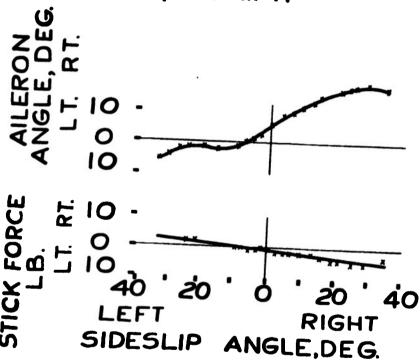
LEFT SIDESLIP ANGLE, DEG.

figure 28.

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UNSATISFACTORY STICK FREE DIHEDRAL EFFECT

FIGHTER AIRPLANE
CLEAN CONDITION, RATED POWER
Vi=110 MPH

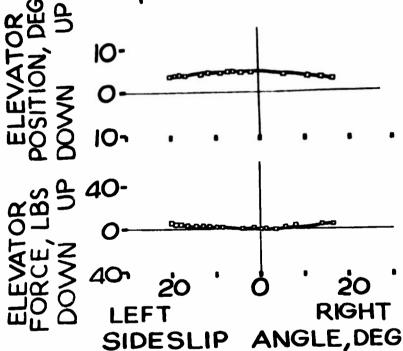


NATIONAL ADVISORY
COMMITTEE FOR AERONAUTICS LMAL 403

Figure 29.

SATISFACTORY PITCHING MOMENT DUE TO SIDESLIP

FIGHTER AIRPLANE
LANDING CONDITION
Vi = 104 MPH



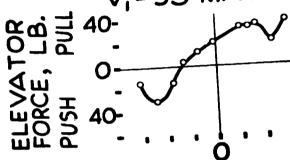
COMMITTEE FOR AERONAUTICS

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Figure 30.

UNSATISFACTORY PITCHING MOMENT DUE TO SIDESLIP

NAVY TORPEDO BOMBER CLEAN, RATED POWER Vi=95 MPH



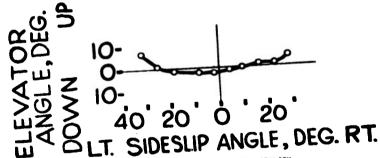


Figure 31.

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(ななな) 日本3日間 20896 DIVISION: Aerodynamics (2) ORIG. AGENCY NUMBER SECTION: Stability and Control (1) MR 14126 CROSS REFERENCES: Airplanes - Control characteristics (08393): Airplanes - Stability (08487): Airplanes. PEVISION security Bomber - Performance charts (08517) AUTHOR(S) AMER. WILE: Charts showing stability and control characteristics of airplanes in flight WIK. FORG'N, TITLE, ORIGINATING AGENCY: National Advisory Committee for Aeronautics, Washington, D. C. TRANSLATION: COUNTRY | LANGUAGE FORG'N CLASS U. S.CLASS. | DATE PAGES ILLUS FEATURES U.S. Unclass Dec 'lll graphs Eng. ΔΩΩΥΩΔΕΥ A collection of charts is presented which indicates the satisfactory and unsatisfactory stability and control characteristics of fighters, Navy scout and patrol bombers and of trainees. Elevator, aileron and rudder control characteristics and dihedral characteristics are also evaluated. These charts were used during s conference to portray good and undesirable airplane flight characteristics. NOTE: Requests for copies of this report must be addressed to: N.A.C.A.. Washington, D. C. 2 T-2, HQ., AIR MATERIEL COMMAND ğ

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